



## Path Planning and Collision Avoidance Algorithms for Small RPA

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### Abstract

The development of Remotely Piloted Aircrafts (RPAS) for civil applications has been rapidly growing over the past years. This work presents a solution for the generation of optimal trajectories for RPAS subject to manoeuvrability and collision avoidance constraints. To achieve this task a two-layered approach is proposed. In the first stage, classical path planning techniques are implemented to generate safe and flyable paths in a known static environment. The A\* algorithm and Ant Colony Optimization (ACO) are used to find an optimal sequence of waypoints in a discrete environment. To ensure that the path is flyable and complies with curvature constraints, an optimization of Rational Bezier curves is implemented. The second stage is developed for real-time implementation and potential fields methods are used to replan the initial path when new obstacles are detected. For the global path planning stage the best results were found to be provided by using ACO to optimize waypoint order, A\* to connect the waypoints and rational Bezier curves with constraint restriction. The Potential Fields method is computationally inexpensive proving to be a feasible solution for real-time implementation. It is shown that the algorithms perform reasonably well in several scenarios.

### Keywords

RPAS; Collision avoidance; Path planning; A\* algorithm; A\* algorithm

# Path Planning and Collision Avoidance Algorithms for Small RPA

## 1 Introduction

Safety is the most important factor in aviation. Remotely Piloted Aircraft (RPA) can present a hazard to other aircraft, people and property owing to their characteristics and specific operational applications. Current civil applications for RPA include infrastructures and traffic monitoring, search and rescue operations among many others. These tasks require that the RPA can autonomously go through specific waypoints while avoiding collisions on the way. Path planning with obstacle avoidance is a fundamental aspect of autonomous vehicles operations. To this day, several solutions have been developed to tackle this problem. The proposed methods are usually divided into two main categories: global and local path planning. Global path planning requires a known static environment and is generally performed offline before the mission begins. Local path planning methods are implemented during mission execution and are responsible for the replanning of the original path when new obstacles are detected.

Graph search algorithms are one of the most popular methods used in robot path planning. These methods are heavily based on the Dijkstra's algorithm [1]: starting at one vertex, a graph is searched by exploring adjacent nodes until the goal state is reached, with the intent of finding the optimal path. In [2], a variation of the A\* algorithm is proposed for path planning of fixed-wing RPAs in 3D environments, providing a feasible solution for offline path planning with turning and climbing angles constraints. Rapidly Exploring Random Tree (RRT) is a popular search algorithm when dealing with high dimensional spaces. In [3], a greedy version of closed-loop RRT is used to plan the collision avoidance path. The collision with manned aircraft is predicted based on the RPA current flight route and the aircraft ADS-B data. The Ant Colony Optimization algorithm has also been applied to the RPA global path planning problem [4],[5]. The solutions, however, are only applied to 2D environments, considering a constant flying height, which is not suitable for many applications of flying vehicles. The Artificial Potential Field methods are an approach inspired by physical potential fields. These methods are generally used for reactive collision avoidance systems [6] and are a good solution for online implementation. In [7], this approach was applied to formation flights. Velocity Obstacles are another approach for local path planning. This method was initially developed for ground vehicles but have since been applied to RPAs [8],[9],[10]. The method also allows cooperative manoeuvres [11].

This work presents a two-stage path planning architecture. In the first stage the global planning module, which assumes a known static environment, determines a collision free path from a given start to goal configurations. This path is given as a reference for the mission execution stage and as new threats are detected by the on-board sensors, the local planning module must replan the path to avoid these new obstacles.

## 2 Path Planning Framework

Figure 1 illustrates the proposed framework for the path planning system and its integration with the other system modules. The navigation module is responsible for the estimation of the RPA state, which comprises its position and velocity. The obstacle detection module contains the sensors and algorithms necessary to detect and estimate the obstacles state. In this work, the type of sensors used will not be specified, but it is assumed that there is a working method of sensor fusion to obtain the necessary information about the environment. For the purpose of collision avoidance, a safety volume is defined around the obstacles. Due to its simplicity and ability to encompass a wide variety of obstacle types a cylindrical model is used to represent obstacles. The pre-flight path planning module is used offline to find an optimal path. During mission execution the planned path  $P_{ref}$ , is given as a reference to the path tracking or path following module that, in conjunction with low level controllers, has the task of finding the necessary control inputs for the RPA to follow the given path. If new obstacles are detected during mission execution, the path replanning module is activated and an avoidance segment  $P_{av}$  is planned.

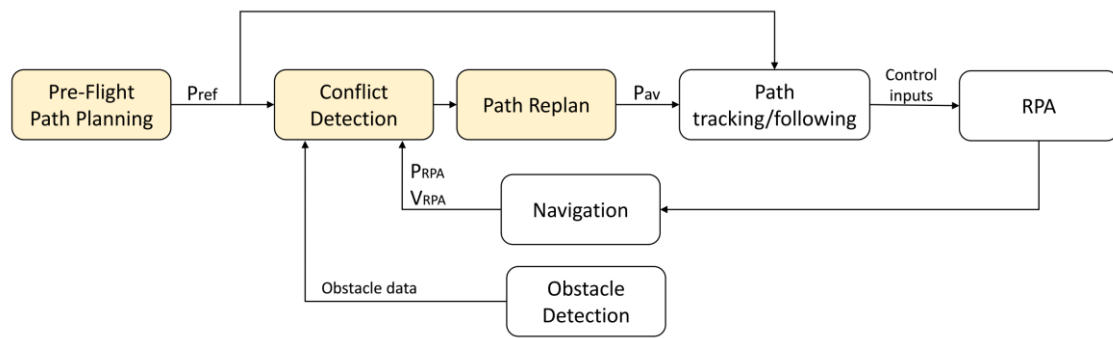


Figure 1 - System architecture

### 3 Pre-Flight Path Planning

The fundamental problem of path planning consists of finding a sequence of actions for an agent that can take it from one location to another while avoiding any obstacles on the way.

#### 3.1 Configuration Space

A key concept of path planning is the representation of the physical world where the RPAS will operate. The environment model includes several natural conditions such as terrain, weather and obstacles.

In this work the configuration space will be defined as a regular grid. This is a conceptually simple representation, easy to construct and by properly defining the grid resolution it is possible to find cinematically feasible paths. This representation is also convenient as an action space can be independently built and a set of common actions that can be applied to any of the states in the configuration space. When deciding on the grid size some limitations of the RPA must be considered. Looking at Figure 2, it is possible to deduce the grid resolution

$$\Delta x = \Delta y = \frac{R_{curv}}{\sqrt{2}} \quad (1)$$

where  $R_{curv}$  is the minimum turning radius that the RPA can perform.

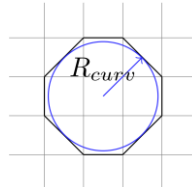


Figure 2 - Grid resolution

Fixed-wing platforms are not allowed to climb at an angle superior to the maximum climb angle,  $\gamma_{max}$ , hence the resolution along the vertical plane is defined according to this limit as

$$\Delta z = \Delta x \times \tan^{-1}(\gamma_{max}) \quad (2)$$

#### 3.2 Constraints

Some of the kinematic constraints of the vehicle, like minimum turning radius and maximum climb angle, were already included in the definition of the search space. Other constraints in the vehicle's manoeuvrability can be included in the process of node expansion during the search process through the graph.

Distinct expansion rules are defined for multicopter and fixed-wing platforms. For a non-holonomic vehicle, or multicopter platforms, any of the 26 neighbouring nodes in a regular grid can be reached as illustrated in Figure 3 (left). Fixed-wing platforms have a forward only motion and cannot make sharp turns or climbs. To incorporate manoeuvrability restrictions, a set of expansion rules is defined as seen in Figure 3 (right).

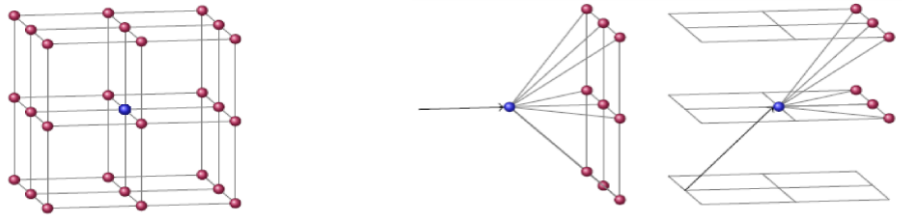


Figure 3 - Expansion rules for multirotor (left) and fixed-wing aircraft (right)

Other constraints are the minimum safety distance. Given the relative distance between the RPA position and the obstacle centre  $d_o$ , the collision avoidance constraint is

$$d_o = \|P_{RPAS} - P_{Obs}\| \geq R_s = R_{obstacle} + d_{min} \quad (3)$$

where  $R_s$  is the obstacle radius plus the minimum allowed distance between vehicle and obstacles and the minimum distance  $d_{min}$ , is defined by considering possible deviations that may occur during the execution of the path.

The mission constraints are the waypoints given to the path planner which the RPA must visit given as

$$WPS = \{P_1, \dots, P_n\}, \quad P_i = [x_i, y_i, z_i] \quad (4)$$

### 3.3 Cost Function

Depending on the mission objectives, different cost functions can be considered. RPAs have limited range and endurance so when planning paths a broadly used criteria is the minimum distance. Looking at the problem of limited on-board energy another important objective would be to plan for least energy cost paths.

#### 3.3.1 Minimum Distance

For the minimum distance paths, the cost function is simply given by the sum of Euclidean distance between all points. Considering a path  $P = \{P_1 \dots P_N\}$  of  $N$  waypoints, the cost is given by

$$F_d = \sum_{i=1}^{N-1} \|P_{i+1} - P_i\| \quad (5)$$

#### 3.3.2 Minimum Energy

To formulate the energy minimization problem, an energy balance is considered. Considering a point mass model for the RPA, its motion can be analysed using the work and energy method. The energy balance is a statement about how much energy is spent when the RPA moves from point  $i$  to point  $j$

$$E_{i \rightarrow j} = \frac{1}{2} m (v_j^2 - v_i^2) + D \Delta s + mg \Delta h \quad (6)$$

where  $m$  is the RPA mass,  $v_i$  and  $v_j$  the vehicle airspeed at points  $i$  and  $j$ ,  $D$  the drag component,  $\Delta s$  the air displacement,  $g$  the gravity acceleration and  $\Delta h$  the height variation between the two points. The cost function for minimum energy paths is then given by

$$F_e = \sum_{i=1}^{N-1} E_{i \rightarrow i+1} \quad (7)$$

This simplified model can be applied to either fixed-wing platforms or multirotors. In the latter case the drag component tends to be negligible.

### 3.4 Path Search

To generate an optimal path for the RPA, two algorithms are considered: A\* and ACO.

### 3.4.1 A\* Algorithm

The A\* algorithm [2] works by systematically searching the graph by applying the transition function and choosing the states that minimize the cost function, given by

$$f(n) = g(n) + h(n) \quad (8)$$

where  $g(n)$  denotes the cost to reach the node and  $h(n)$  represents the cost of getting from the node to the goal, while keeping track of the visited nodes so that no redundant exploration occurs.

This method is known to be complete (it always finds a solution if one exists) and optimal (the solution found is the optimal one) if the heuristic function is admissible (it never overestimates the solution cost) and consistent (for every node  $n$  and every successor  $n'$  the cost of reaching the goal from  $n$  is less than the step cost from  $n$  to  $n'$  plus the cost from  $n'$  to the goal).

### 3.4.2 Ant Colony Optimization

Ant Colony Optimization [4],[5] is a metaheuristic method derived from the observation of real ant's behaviour that use a pheromone trail to mark paths from the nest to the food source. A set of ants is placed at the departure node and following a probabilistic model they transition between nodes until all the required waypoints have been visited. Once each ant has found a solution, the pheromone trail is updated giving more emphasis to the best solution found so far. While ants construct their solution the transition probability of the  $k$  ant move from node  $i$  to node  $j$  is given by a random proportional rule

$$p_{ij}^k = \frac{[\tau_{ij}]^\alpha \cdot [\eta_{ij}]^\beta}{\sum [\tau_{ij}]^\alpha \cdot [\eta_{ij}]^\beta} \quad (9)$$

where  $n_{ij}$  is the heuristic value,  $\tau_{ij}$  the pheromone value and  $\alpha$  and  $\beta$  determine the influence of the pheromone trail and the heuristic information.

**Pheromone trail:** the pheromone trail represents the desirability of visiting one node after the other. Generally pheromones are deposited in the edges connecting the graph nodes, however when planning in a large tri-dimensional grid it is infeasible to define each possible edge connecting nodes so in this implementation pheromones will be deposited in each node instead of the edges.

**Heuristic information:** to ensure that ants reach the target point, the heuristic value is defined either as a measure of distance or a measure of energy expenditure. The heuristic information is computed according to

$$\eta_{ij} = \left(\frac{1}{d_{ij}}\right)^c \left(\frac{1}{e_{ij}}\right)^{1-c} \quad (10)$$

where  $d_{ij}$  represents the Euclidean distance between node  $i$  and node  $j$  and  $e_{ij}$  represents the energy spent in the transition between nodes, and calculated using (6). The constant  $c$  is 1 when distance is minimized and zero if energy is minimized.

### 3.5 Path Smoothing

The paths obtained with A\* and ACO consist of straight-line segments between waypoints. These paths cannot be exactly followed by a RPA with dynamic and kinematic constraints. Bezier curves are used to generate a flyable path for the RPA. Bezier curves are a type of parametric curves designed to provide a smooth path that passes exactly through the initial and final waypoints and is influenced by the other waypoints on the way, which are defined as control points. A particular case of these curves are Rational Bezier curves [12]. These curves are generated by attributing a weight to each control point, pulling or pushing the curve away from the point. They allow a better control over the curve shape. These curves are given by

$$P_R(t) = \frac{\sum_{i=0}^n B_i^n(t) w_i P_i}{\sum_{i=0}^n B_i^n(t) w_i} \quad t \in [0,1] \quad (11)$$

$$B_i^n(t) = \binom{n}{i} (1-t)^{n-1} t^i, \quad i \in \{0, 1, \dots, n\} \quad (12)$$

where  $B_i(t)$  is the Bernstein polynomial,  $P_i$  the control points given, by  $A^*$  and ACO, and  $w_i$  the curve weights. The curvature of a parametric curve  $P(t)$  can be calculated as

$$\kappa(t) = \frac{|P'(t) \times P''(t)|}{|P'(t)|^3} \quad (13)$$

The defined problem is to optimize the weights of a rational Bezier curve. The optimization problem is formulated as

$$\text{minimize } F_c \quad (14)$$

$$\text{subject to } d_o \geq R_s \quad (15)$$

$$P_R(t) = f(w) \quad (16)$$

$$|k| \leq k_{max} \quad (17)$$

$$w_{min} \leq w_i \leq w_{max} \quad (18)$$

The cost function  $F_c$  to be minimized is given either by eqs. (5) or (6). If the constraints are to be satisfied without optimizing the cost function  $F_c$  is set to zero. The constraint defined by eq. (15) imposes a minimum distance between the RPA and the obstacle, and eq. (17) ensures that, given the RPA turning limits, the path is flyable.

A generic constrained optimization solver, *fmincon*, provided by MATLAB is used to solve the problem. The safety distance and curvature are calculated for each point on the Bezier curve, but the number of points on the curve differ from the number of optimization variables, so a constraints lumping method is performed to attribute to each optimization variable the maximum constraint value of the closest point.

High degree curves are generally not efficient to process and situations where a solution cannot be found can easily arise when planning a long range mission on an area densely populated with obstacles. To solve this issue the curve is successively divided and each segment is optimized until the constraints are satisfied. Algorithm 1 describes the overall pre-flight path planning method.

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**Algorithm 1: Pre-flight path planning**

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**Input:** Constraints, cost function, reference waypoints, obstacles and departure heading and flight path angle;

**Output:** Optimal path from start to goal,  $P_{op}$ ;

Find control points  $CPs = \{P_1, \dots, P_N\}$ , using graph search ( $A^*/ACO$ );

Set unitary weights and calculate initial Bezier curve,  $B_i$ , using eq. (11);

Current curve  $\leftarrow B_i$ ;

**while** Solution is not found **do**

Find optimal weights,  $w_o$ , for the current Bezier curve;

Current curve  $\leftarrow P_R(w_o)$ ;

**if** constraints are satisfied **then**

$B_o \leftarrow$  Current curve;

**return**  $B_o$

**else**

Divide current curve

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## 4 Real-Time Path Planning

This section addresses the problem of replanning a reference path when new obstacles are detected, while taking into account the Rules of the Air.

To develop the module for real-time path planning, some safety distances are defined. First, a detection radius  $R_d$  defines the distance at which the obstacle is acknowledged by the path replanning system. The action radius  $R_a$  defines the distance from which the replanned path begins to depart from the original path given by the global planner. The safety radius  $R_s$  defines the required safety distance that must be maintained. A collision is said to occur when the obstacle breaches the collision radius  $R_c$ . These distances are illustrated in Figure 4.

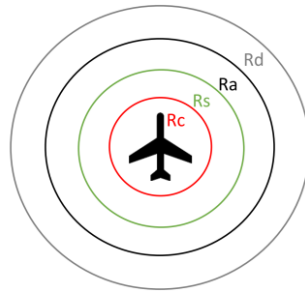


Figure 4 - Safety radii defined around the aircraft

#### 4.1 Rules of the Air

Regarding the avoidance of collisions for manned flight, four main points are stated [13]:

1. On a head-on encounter, both aircraft should deviate to the right;
2. On a converging scenario, the aircraft with the other on its right-hand side has to give way and turn right;
3. In an overtaking event, the faster aircraft must overcome on the right hand side of the slower one;
4. An aircraft should avoid passing over, under or in front of other.

To comply with the Rules of the Air when a moving obstacle is detected, the type of encounter must be evaluated. Depending on the type of encounter, different resolutions are adopted:

- When the intruder is found to be in a head-on collision course or to the right of the RPA, the avoidance should be made by turning right;
- If the intruder is approaching from the left, and the RPA is in level flight, turning right will put the RPA in front of the intruder. To avoid this scenario, the avoidance manoeuvre is made by turning left and going behind it;
- If the RPA is climbing, the avoidance is made by levelling the flight until the intruder is overcome;
- If the RPA is descending, the aircraft could be levelled off, but due to inertia this would be riskier than increasing the descent rate (unless the value is at its maximum).

If a static obstacle appears in the way, different paths are achieved depending on the direction of the avoidance. In this situation, there are no rules commanding the vehicle to behave a certain way, so the following strategy is adopted:

- If any side of the obstacle is blocked, the rotation is set to the opposite direction;
- Considering the line joining the RPA position and the obstacle centre, if the goal point in the path is to the left the rotation is made counter clockwise and vice versa. If the point or path direction is along the line, the swirl direction can be arbitrarily chosen.

#### 4.2 Collision Detection

To detect possible collisions, the concept of the Closest Point of Approach (CPA) is used. When a conflict with multiple intruders occurs, threats must be prioritized. As intruders will have different speeds and bearings, using the distances to the collision point is not enough, so the time to collision,  $t_{CPA}$ , is used instead. Higher priority will be given to intruders with the smallest  $t_{CPA}$  and the conflicts are resolved in a sequential manner.

#### 4.3 Potential Fields

In this approach the obstacles and the goal position are treated as charged particles [6]. A repulsive force is attributed to the obstacles and an attractive force to the goal point. The sum of those forces is used to generate the direction of motion. The proposed fields are generated in a similar way to [14],[15].

The attractive potential is responsible for directing the RPA towards the desired destination. If the objective is to direct the vehicle to a single goal waypoint, the potential function

$$F_{at} = \frac{P_{WP} - P}{|P_{WP} - P|} \quad (19)$$

is simply given by the direction from the current position,  $P$ , to the desired waypoint,  $P_{WP}$ . This potential is depicted in Figure 5 (left).

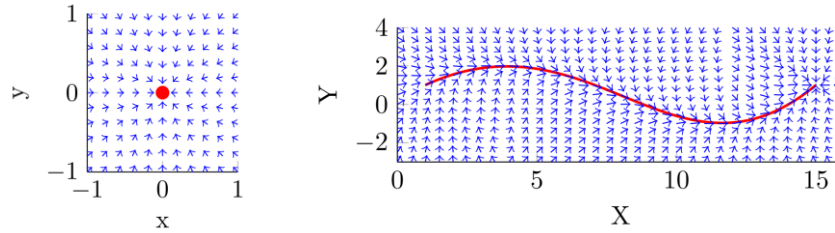


Figure 5 - Attractive potential field for a waypoint (left) and for a path with  $\alpha = 0.5$  (right)

When the mission consists on following a pre-planned path, the potential function must take into account two terms: one that brings the RPA close to the given path and other that makes the vehicle follow the path direction. To obtain the first term, the closest point on the path,  $P_{close}$ , to the current position,  $P$ , is found and the direction between both is taken. The path following term is obtained from the direction from the closest point on the path to the next point on the path,  $P_{next}$ , resulting in the potential function:

$$F_{at} = \alpha_{PF} \frac{P_{close} - P}{\|P_{close} - P\|} + (1 - \alpha_{PF}) \frac{P_{next} - P_{close}}{\|P_{next} - P_{close}\|} \quad (20)$$

By selecting the values of  $\alpha_{PF}$ , more importance can be given to the path following or the path approaching direction. This potential field can be seen in Figure 5 (right).

The repulsive force, which keeps the vehicle away from obstacles, is given by

$$F_{rep} = \begin{cases} 0, & \text{if } d_o \geq R_a \text{ or } ang \geq \theta_{cut} \\ -\frac{d_o}{\|d_o\|} \left| \frac{R_t - d_o}{R_t} \right| S, & \text{if } R_c \leq d_o \leq R_a \\ \infty, & \text{if } d_o \leq R_c \end{cases} \quad (21)$$

If the distance to the obstacle is greater than the action radius, the obstacle has no influence and the potential is zero. For distances inferior to the collision radius, the potential is infinite and points in the opposite direction of the vector connecting the current point to the obstacle centre. Between the action and collision radius, the potential is dependent on three terms: the first one keeps the RPA at a distance and depends on the distance to the obstacle centre, the second term increases the field intensity as the vehicle gets closer to the obstacle and the last term induces a swirling motion to provide a smooth movement.

A cut-off angle,  $\theta_{cut}$ , is defined to reduce the repulsive potential once the obstacle is overcome to prevent the RPA from being trapped around the obstacle. The angle between the desired direction of motion,  $D$ , and the relative position between the RPA and the obstacle is given by

$$ang = \frac{\cos^{-1}(Dd_o)}{\|D\| \|d_o\|} \quad (22)$$

To comply with the avoidance logic, the swirling direction,  $S$ , is defined according to the type of encounter. An example of the repulsive field can be seen in Figure 6.

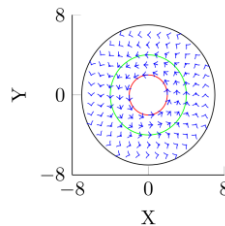


Figure 6 - Repulsive potential flow

The total potential flow force, which determines the movement direction, is given by



$$\mathbf{F}_{tot} = \mathbf{F}_{at} + \sum \mathbf{F}_{rep} \quad (23)$$

From the total field vector the required heading and flight path angles to avoid the obstacle are obtained, from which, knowing the current direction of motion, a series of waypoints are generated until the obstacle has been cleared.

However, the combination of both the attractive and repulsive potential can lead to heading changes not feasible by the RPA, so the angle between the platform current heading and  $\mathbf{F}_{tot}$  is taken. If this angle is greater than the maximum turning angle, the angle is scaled to the maximum allowable value. The same applies to the climb angle.

One issue may arise when an intruder obstructs one of the required waypoints defined during the mission planning stage. If the obstacle is static there is no way to go through the required waypoint without violating the safety distance, however if the obstacle is moving it is possible to return to the required waypoint once the collision has been avoided. To do so instead of returning to the global path once the threat is overcome, the attractive potential function for a waypoint is activated and a path that directs the RPA towards the missed waypoint is computed. When the waypoint has been passed over, or the RPA has come within a predefined distance, the RPA returns to the global path.

## 5 Results

All examples were obtained with MATLAB R2016a running on an Intel Core i5 with a CPU of 2.4 GHz, 4Gb RAM and Windows 7.

### 5.1 Pre-Flight Path Planning

For the pre-flight path planning stage, an example for minimum distance paths between three waypoints is presented. The following parameters are used for the ACO algorithm:  $\alpha = 1$ ,  $\beta = 2$ , number of ants  $N = 10$ ,  $q_0 = 0.9$ ,  $\tau_0 = 1$ ,  $\rho = 0.3$ ,  $\eta = 0.9$ , 500 iterations,  $\tau_{min} = 0.1$  and  $\tau_{max} = 10$ . The results are presented for a fixed-wing RPA with the following parameters: airspeed  $v = 16\text{m/s}$ , mass  $m = 2\text{kg}$ , wing area  $S = 1.5\text{m}^2$ , minimum turning radius  $R_{curv} = 10\text{m}$  and maximum climb angle  $\theta_{max} = 30\text{deg}$ .

In the following example, three waypoints are considered. In Figure 7(a), a random waypoint order is given to the A\* algorithm, while in Figure 7(b), the ACO is used to find the optimal waypoint order and the A\* algorithm is used to connect the waypoints. When comparing the results of using only the A\* algorithm (Table 1(a)), and using a combination of A\* and ACO (Table 1(b)), it is concluded that the offline planner provides the best results when A\* is used to plan between waypoints and ACO used to optimize waypoint order.

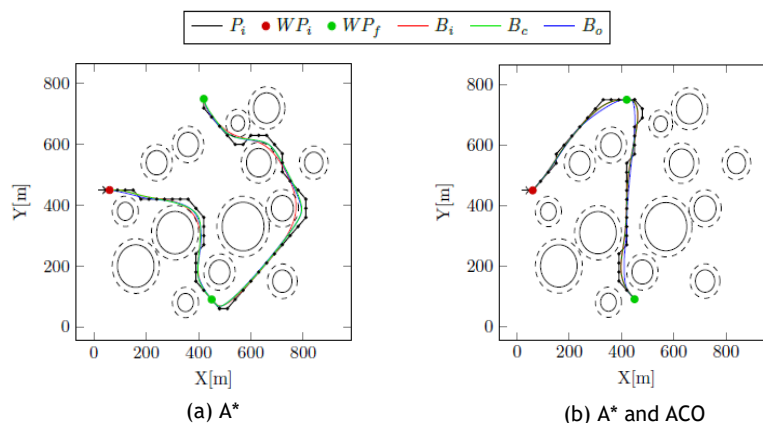


Figure 7 - Three waypoints minimum distance paths results

Table 1 - Results for 3 waypoints example

(a) Minimum distance paths with A*			(b) Minimum distance paths with A* and ACO		
Path	Distance [km]	CPU time [s]	Path	Distance [km]	CPU time [s]
A*	1.885	0.110	A* and ACO	1.266	110.15
$B_i$	1.666	0.052	$B_i$	1.191	0.04
$B_c$	1.703	58.97	$B_c$	1.191	15.60
$B_o$	1.701	212.30	$B_o$	1.175	198.56

## 5.2 Path Replanning

This section presents an example where a segment of the original path must be replanned to avoid new obstacles, two moving and one static. In this case, the potential fields approach was used with the following parameters:  $R_d = 50\text{m}$ ,  $\alpha_{PF} = 0.5$ ,  $\theta_{cut} = 30\text{deg}$ . It is assumed that any detected moving obstacle will maintain its course of motion. In this example, the RPA encounters two moving intruders, with one of them blocking a reference waypoint, and a static obstacle while following the reference path. The resulting replanned path is depicted Figure 8.

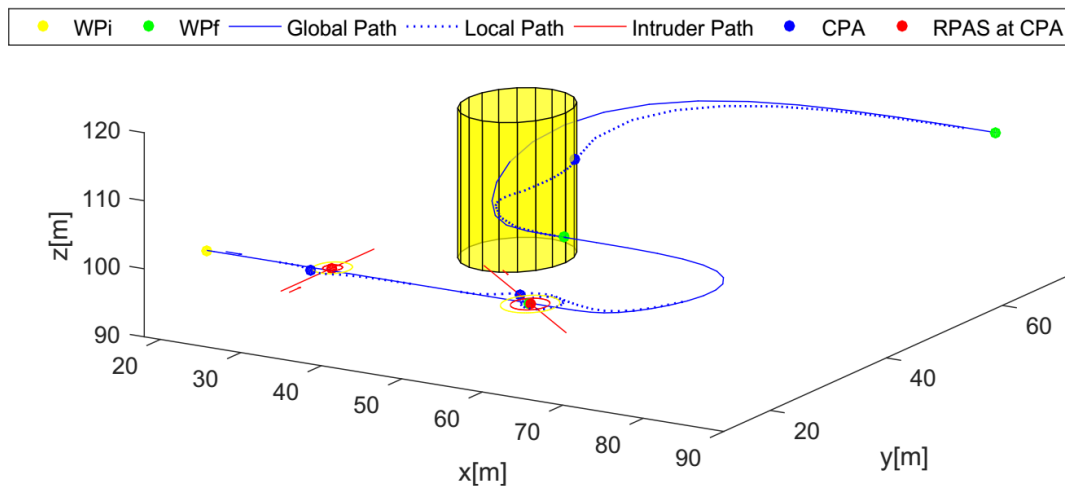


Figure 8 - 3D view of path with multiple encounters

The RPA successfully avoids the collisions but, as seen in Table 2, in the case of the second obstacle, even though no collision occurs, the safety distance is not maintained. When avoiding the static obstacle, the path is replanned to maintain minimum deviation from the initial path.

Table 2 - Replanned results for new static and dynamic obstacles

Obstacle	Minimum distance to obstacles [m]	Safety distance [m]
1	2.32	2
2	2.89	3
3	6.19	6

## 6 Conclusions

This work was developed with the aim of investigating and implementing methods that can provide autonomous flight capabilities to RPAS with collision avoidance capabilities.

The task was divided into a global and a local layer. For the global path planning stage the best results were found to be provided by a combination of the two algorithms, using A\* and ACO to optimize waypoint order. Regarding the Bezier curves, optimizing the cost function provided the minimum cost paths, but the improvements over a curve calculated to meet only the safety and curvature constraints were not significant when considering the increase in computational time. For the online stage Potential Fields were used to generate a local trajectory when

unknown obstacles are detected. The replanning of the path is made considering an uncooperative situation between the vehicles, and a sequential resolution of encounters, prioritized according to time to collision.

The global planner can resolve a series of different scenarios and new optimization criteria can be easily added to expand the range of problems to solve. The Potential Fields method is computationally inexpensive being a feasible solution for real-time implementation. The algorithm was developed to provide a path to a waypoint manager that guides the RPA through the given list, but it can easily be adapted to be integrated with the lower level control modules serving as a navigation system for a reference motion.

To tackle increasingly complex scenarios some issues still need to be addressed. A better integration of the vehicle dynamics is important to improve the system reliability and performance. A complete solution should consider a cooperative scenario where the vehicles exchange flight plans among each other. The type of sensors used must also be taken into account as they are a crucial part of the real-world implementation that can significantly affect the performance of the system. Different obstacle configurations must be incorporated to encompass the diversity found in the real world.

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